

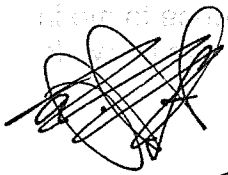
$$\int \frac{M \cdot \bar{m} \, dx}{EI} = \int \left(\frac{M}{EI} \right) \bar{m} \, dx$$

any function linear (always)

$$\int f(x)(ax+b) \, dx =$$

↳ whatever.

$$a \int f(x) \cdot x \, dx + b \int f(x) \, dx =$$



$$\left[\frac{a \int f(x) \cdot x \cdot dx}{\int f(x) \, dx} + b \right] \cdot \int f(x) \, dx$$

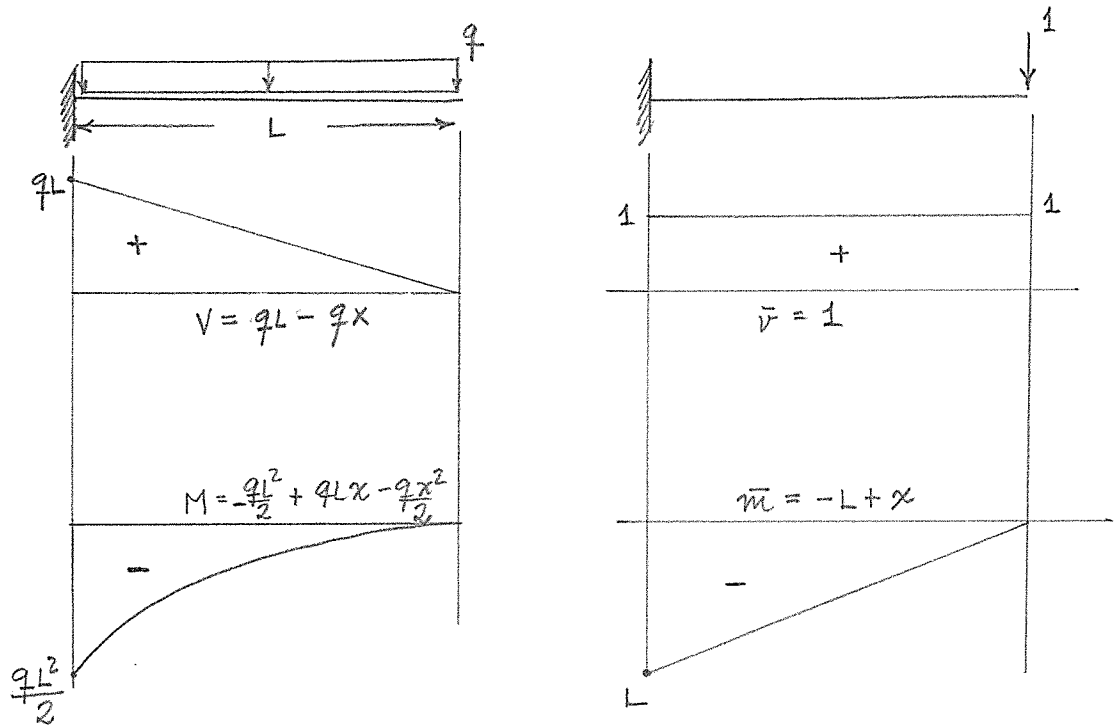
centroid
of $f(x)$.

$$[a \cdot x_c + b] \cdot A_f = \text{scribbled out}$$

\bar{m} @ centroid
of $\frac{M}{EI}$

area under $\frac{M}{EI}$

$$\int \left(\frac{M}{EI} \right) \cdot \bar{m} dx = \left(\int \frac{M}{EI} \right) \cdot \bar{m}(x_c)$$



$$\int_0^L \frac{M \bar{m}}{EI} dx = \frac{1}{EI} \int_0^L \left(-\frac{qL^2}{2} + qLx - \frac{qx^2}{2} \right) (x-L) dx$$

$$= \frac{1}{EI} \int_0^L \left(-\frac{qL^2}{2}x + qLx^2 - \frac{qx^3}{2} + \frac{qL^3}{2} - qL^2x + \frac{qx^2L}{2} \right) dx$$

$$= \left[-\frac{qL^2}{4}x^2 + \frac{qLx^3}{3} - \frac{qx^4}{8} + \frac{qL^3x}{2} - qL^2x^2 + \frac{qLx^3}{6} \right]_0^L \cdot \frac{1}{EI}$$

$$\frac{1}{EI} \left[-\frac{qL^4}{4} + \frac{qL^4}{3} - \frac{qL^4}{8} + \frac{qL^4}{2} - \frac{qL^4}{2} + \frac{qL^4}{6} \right] = \boxed{\frac{qL^4}{8EI} = \Delta_M}$$

$$\int_0^L \frac{V \bar{v}}{KAG} dx = \int_0^L \frac{q(L-x) \cdot 1 \cdot dx}{KAG} = \left[Lx - \frac{x^2}{2} \right]_0^L \cdot \frac{q}{KAG}$$

$$= \boxed{\frac{qL^2}{2KAG} = \Delta_V}$$

FOR RECTANGULAR CROSS SECTION

$$K = \frac{10(1+\nu)}{12+11\nu}$$

FOR CIRCULAR CROSS SECTION

$$K = \frac{6(1+\nu)}{7+6\nu}$$

ν = Poisson's ratio = 0.30 for steel
0.20 " concrete

say we have a rectangular section

$$A = b \cdot h$$

$$G = \frac{E}{2(1+\nu)}$$

$$I = \frac{bh^3}{12}$$

$$\Delta = \Delta_M + \Delta_V = \frac{qL^4}{8E \cdot \frac{1}{12}bh^3} + \frac{qL^2}{2Kbh \cdot \frac{E}{2(1+\nu)}}$$

$$\Delta = \frac{3qL^4}{2bh^3E} + \frac{qL^2 \cdot 2(1+\nu)}{2bhkE} = \frac{qL}{bE} \left[\frac{3}{2} \cdot \left(\frac{L}{h}\right)^3 + \left(\frac{1+\nu}{K}\right) \cdot \left(\frac{L}{h}\right) \right]$$

$$\Delta = \frac{qL}{bE} \left[\underbrace{\frac{3}{2} \left(\frac{L}{h}\right)^3}_{\text{BENDING}} + \overbrace{\frac{12+11\nu}{10} \cdot \left(\frac{L}{h}\right)}^{\text{SHEAR}} \right]$$

Using the fact that:

$$\int_0^L \frac{M\bar{m}}{EI} dx = A_{\frac{M}{EI}} \cdot \bar{m}(\bar{x}_{\frac{M}{EI}})$$

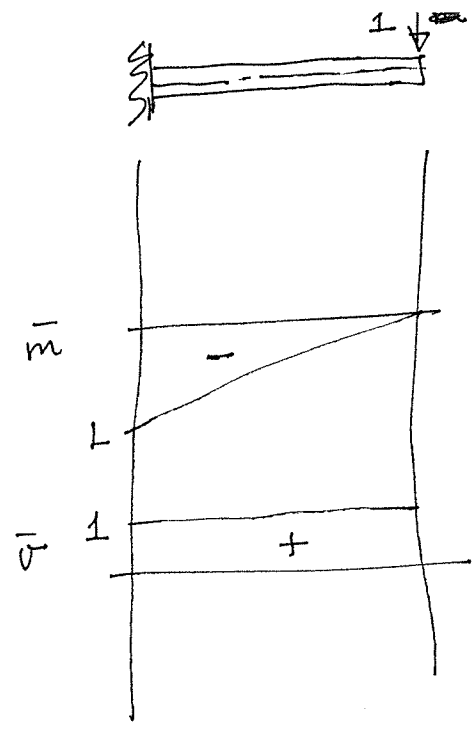
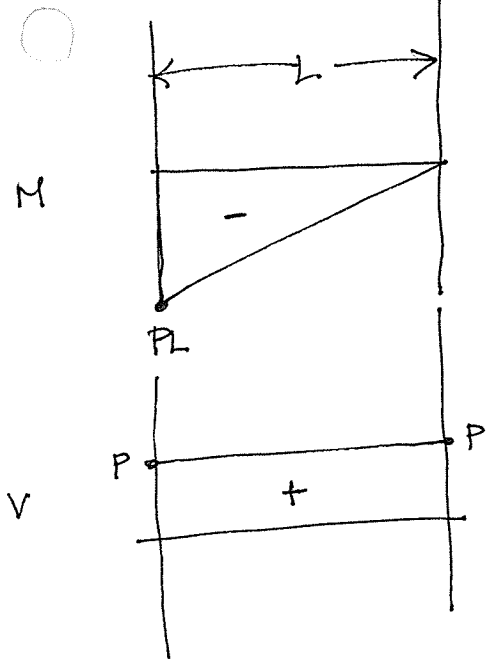
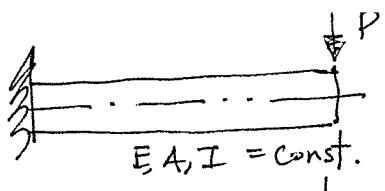
$$A_{\frac{M}{EI}} = \frac{1}{3} \cdot \frac{qL^2}{2} \cdot L = \frac{qL^3}{6EI}$$

$$\bar{x}_{\frac{M}{EI}} = -L + \frac{L}{4} = \frac{3L}{4}$$

$$A_{\frac{M}{EI}} \cdot \bar{m}(\bar{x}_{\frac{M}{EI}}) = \frac{qL^3}{6EI} \cdot \frac{3L}{4} = \frac{qL^4}{8EI}$$

$$\Delta_M = \frac{qL^4}{8EI}$$

$$\int \frac{V\bar{v}}{KAG} dx = A_V \cdot \bar{v}(\bar{x}) = \frac{qL^2}{2} \cdot 1 \cdot \frac{1}{KAG} = \frac{qL^2}{2KAG} = \Delta_V$$



$k = \frac{5}{6}$

 for rectangular sections

$$G = \frac{E}{2(1+\nu)}$$

$$\Delta = \int_0^L \frac{M \bar{m}}{EI} dx + \int_0^L \frac{V \bar{v}}{kAG} dx$$

$$\Delta = \frac{1}{3} \cdot \frac{PL \cdot L \cdot L}{EI} + \frac{6P \cdot 1 \cdot L}{5kAG}$$

$$\Delta = \frac{PL^3}{3EI} + \frac{6PL}{5AG} = \frac{4}{3} \frac{PL^3}{Eb^4} + \frac{6PL \cdot 2(1+\nu)}{5b^2 E}$$

~~Handwritten scribbles and crossed-out equations.~~

For Concrete $\nu \approx 0.17$

$$\frac{PL}{15Eb} \left[20 \left(\frac{L}{b} \right)^3 + 42 \left(\frac{L}{b} \right) \right]$$

for $\frac{L}{b} = 3$	for $\frac{L}{b} = 5$
$\frac{20 \left(\frac{L}{b} \right)^3}{42 \left(\frac{L}{b} \right)} \approx 4.29$	≈ 12

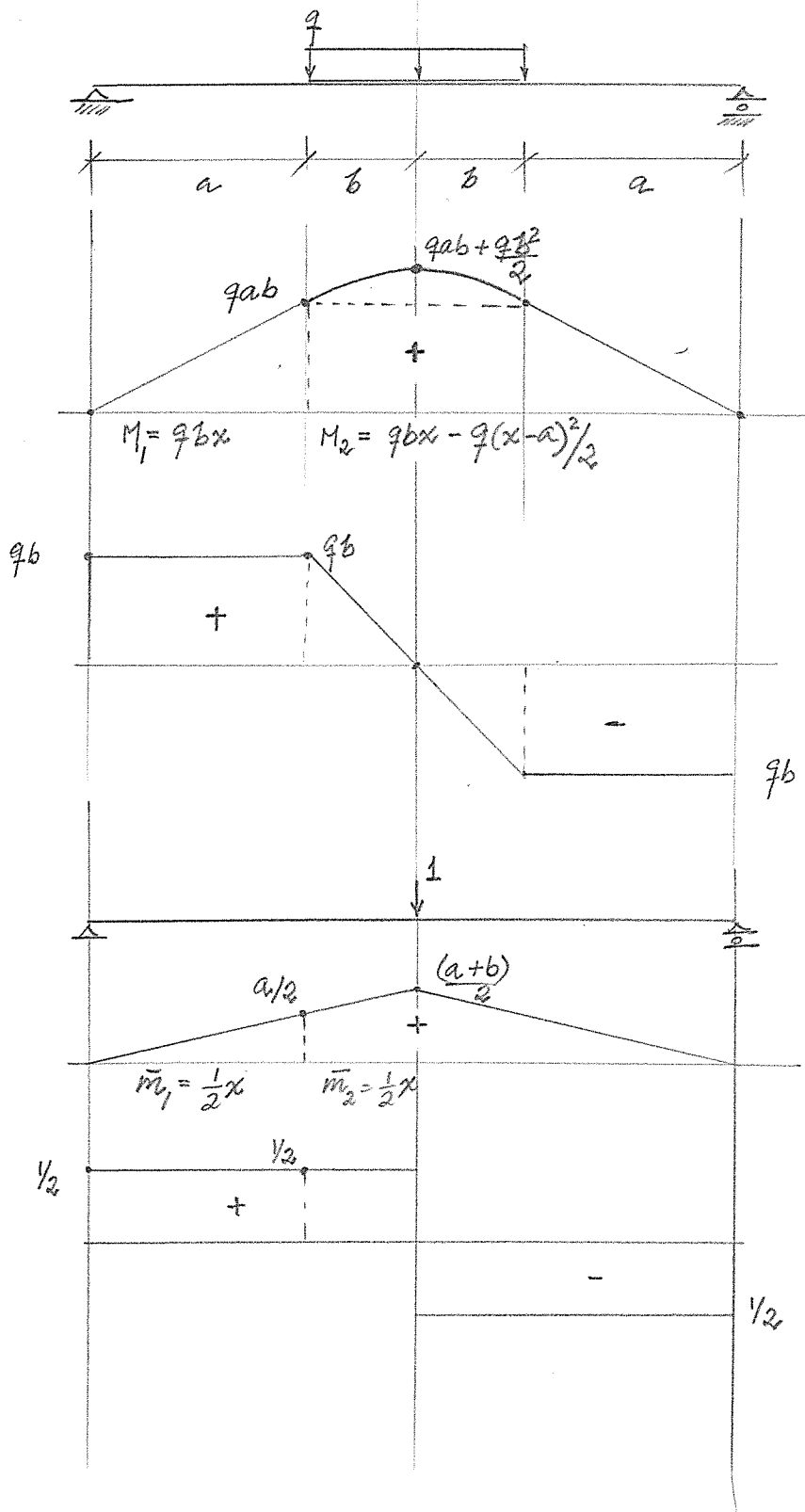
$$\frac{4PL^3}{3Eb^4} + \frac{12PL(1+\nu)}{5Eb^2} =$$

$$\left[20 \left(\frac{L}{b} \right)^3 + 36 \left(\frac{L(1+\nu)}{b} \right) \right] \frac{PL}{15Eb}$$

$$\left[20 \left(\frac{L}{b} \right)^3 + 36(1+\nu) \left(\frac{L}{b} \right) \right] \frac{PL}{15Eb} = \Delta$$

\downarrow SHEAR

Shear deformations can be neglected for $\frac{L}{b} > 5$ less than 10%



$EI = \text{const.}$

$$\int \frac{M \bar{m} dx}{EI} = A_M \bar{m}(\bar{x})$$

for section 1.

$$A_{M_1} = \frac{1}{2} q a b \cdot a = \frac{q a^2 b}{2} \quad \bar{x} = \frac{2a}{3}$$

$$\bar{m}(\bar{x}) = \frac{1}{2} \left(\frac{2a}{3} \right) = \frac{a}{3}$$

$$A_{M_1} \cdot \bar{m}(\bar{x}) = \frac{q a^3 b}{6EI}$$

for section 2.

rect. parabola

$$A_{M_2} = q a b \cdot b + \frac{x}{3} \cdot (q a b + \frac{q b^2}{x} - q a b) \cdot b$$

$$\sum A_{M_{2i}} \cdot m(\bar{x}_i) = q a b^2 \cdot \frac{1}{2} \left(a + \frac{b}{2} \right) + \frac{q b^3}{3} \cdot \frac{1}{2} \left(a + \frac{5}{8} b \right)$$

$$= \frac{q a^2 b^2}{2} + \frac{q a b^3}{4} + \frac{q a b^3}{6} + \frac{5 q b^4}{48}$$

$$= \frac{q a^2 b^2}{2EI} + \frac{5 q a b^3}{12EI} + \frac{5 q b^4}{48EI}$$

$$\Delta = 2 \left[\frac{q a^3 b}{6EI} + \frac{q a^2 b^2}{2EI} + \frac{5 q a b^3}{12EI} + \frac{5 q b^4}{48EI} \right]$$

$$\Delta = \int_0^L \frac{M\bar{u} dx}{EI} + \int_0^L \frac{V\bar{v} dx}{kAG}$$

$$\Delta = \int_0^a \frac{(qb x) \left(\frac{1}{2}x\right) dx}{EI} + \int_a^{b+a} \left(qbx - \frac{q(x-a)^2}{2}\right) \left(\frac{1}{2}x\right) \frac{dx}{EI}$$

$$\Delta = \frac{qb x^3}{6EI} \Big|_0^a + \frac{qb x^3}{6EI} \Big|_a^{b+a} - \left(\frac{qx^4}{16EI} - \frac{qax^3}{6} + \frac{qa^2 x^2}{8} \right) \Big|_a^{b+a}$$

$$\Delta = \frac{qa^3 b}{6EI} + \frac{qb(a+b)^3}{6EI} - \frac{qa^3 b}{6EI}$$

$$- \frac{q(a+b)^4}{16EI} + \frac{qa(a+b)^3}{6EI} - \frac{qa^2(a+b)^2}{8EI}$$

$$\boxed{+ \frac{qa^4}{16EI} - \frac{qa^4}{6EI} + \frac{qa^4}{8EI}}$$

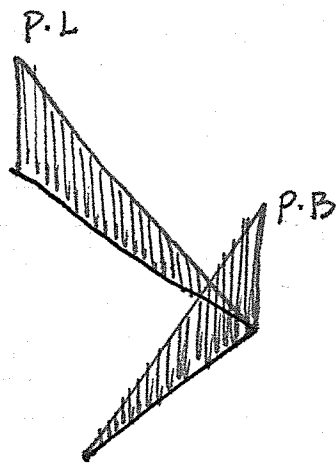
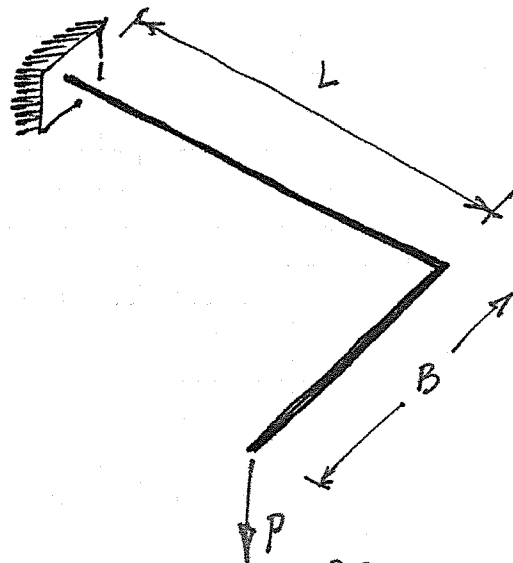
$$\rightarrow \frac{qa^4}{48}$$

$$\Delta = 2 \left[\frac{qb(a+b)^3}{6EI} - \frac{q(a+b)^4}{16EI} + \frac{qa(a+b)^3}{6EI} - \frac{qa^2(a+b)^2}{8EI} + \frac{qa^4}{48EI} \right]$$

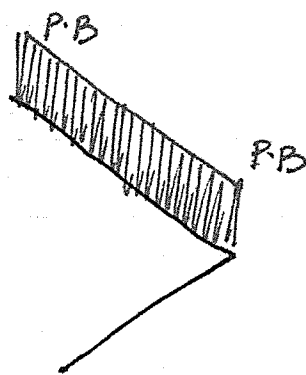
VIRTUAL WORK to Compute deformations in 3-D frame structures.

$$\delta = \int \frac{N\bar{u} dx}{AE} + \int \frac{V\bar{v} dx}{a_r G} + \int \frac{M\bar{m} dx}{EI} + \int \frac{T\bar{\theta} dx}{GJ}$$

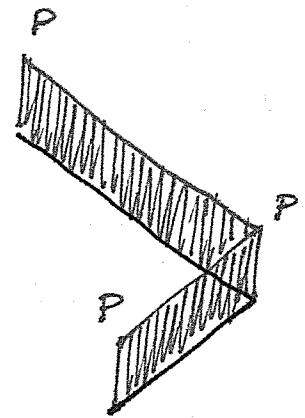
\downarrow \downarrow \downarrow \downarrow
 Axial Shear Bending Moment Torsion



Bending Moment



Torsion



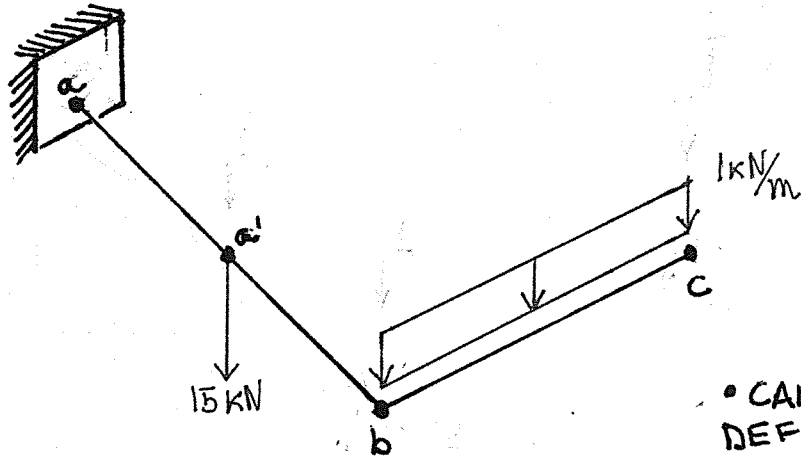
Shear

$$\int \frac{N\bar{u}}{AE} dx = 0$$

$$\int \frac{V\bar{v}}{a_r G} dx = \frac{(P \cdot L)(L)}{a_r G} + \frac{(P \cdot B)(L)}{a_r G}$$

$$\int \frac{M\bar{m}}{EI} dx = \left(\frac{1}{2} PL^2\right) \cdot \left(\frac{2}{3} L\right) \frac{1}{EI} + \left(\frac{1}{2} PB^2\right) \left(\frac{2}{3} B\right) \frac{1}{EI}$$

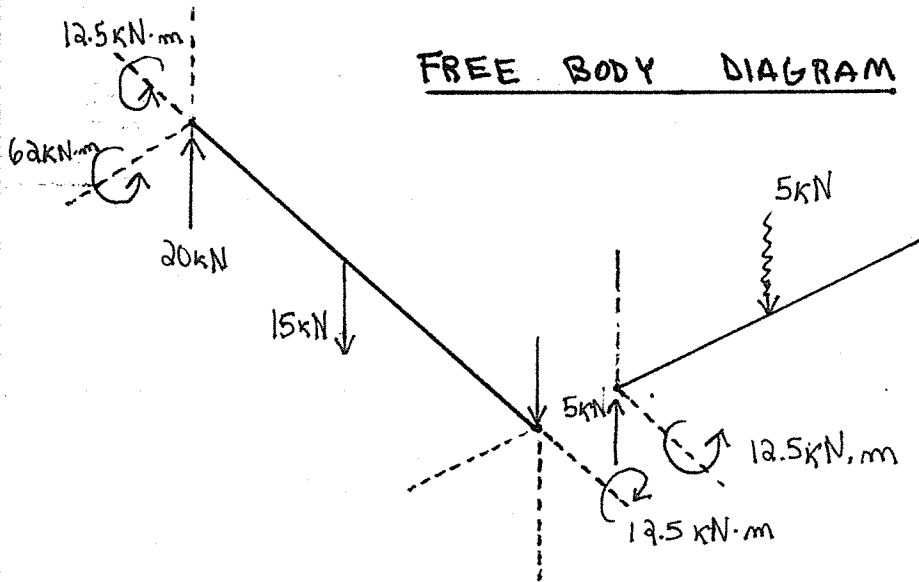
$$\int \frac{T\bar{t}}{GJ} dx = (P \cdot B \cdot L)(B) \frac{1}{GJ}$$



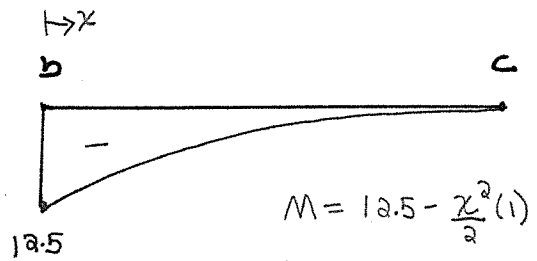
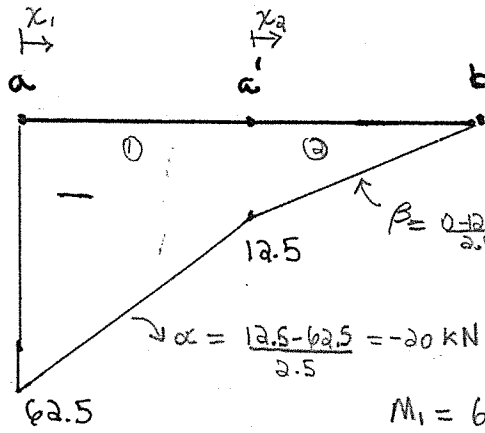
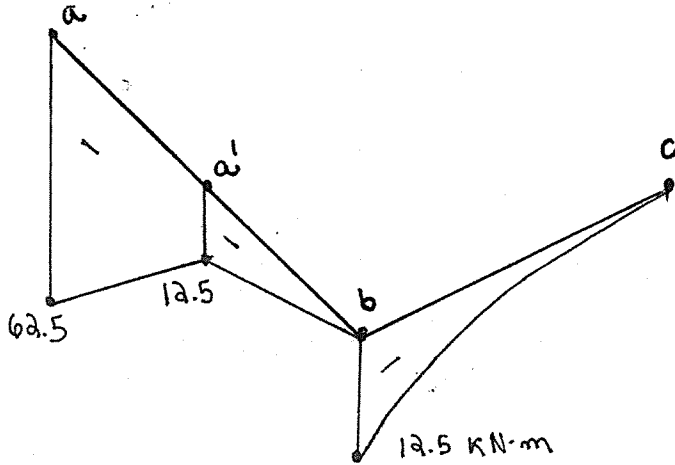
$a-a' = 2.5\text{ m}$
 $a-b = 5.0\text{ m}$
 $b-c = 5.0\text{ m}$

• CALCULATE VERTICAL DEFLECTION @ a', b, c

FREE BODY DIAGRAM



BENDING MOMENT (KN.m)



$$\beta = \frac{0 - 12.5}{2.5} = -5 \text{ KN}$$

$$\alpha = \frac{12.5 - 62.5}{2.5} = -20 \text{ KN}$$

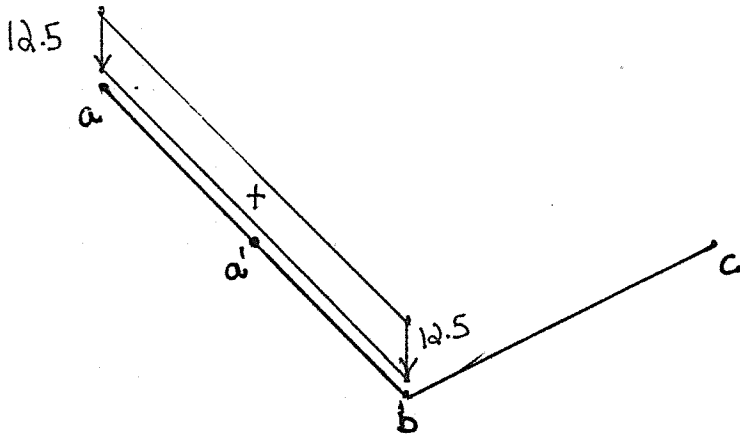
$$M_1 = 62.5 - 20x_1$$

$$0 \leq x_1 \leq 2.5$$

$$M_2 = 12.5 - 5x_2$$

$$0 \leq x_2 \leq 2.5$$

TORSION (KN.m)

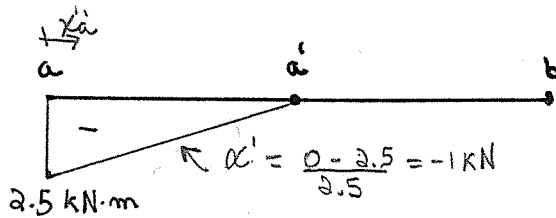


$$T_1 = 12.5 \quad \forall x \text{ a-b}$$

$$T_2 = 0 \quad \forall x \text{ b-c}$$

$$\delta_{vi} = \int_a^c \frac{M \bar{m}}{EI} dx + \int_a^c \frac{T \bar{t}}{GJ} dx$$

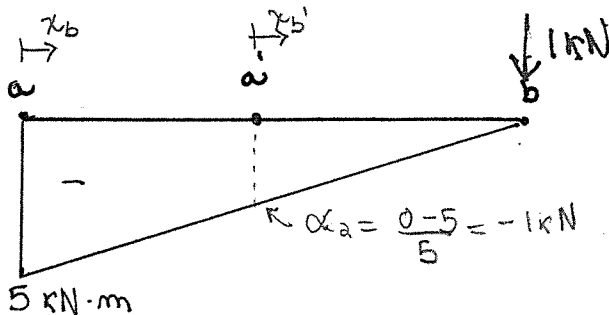
$\delta_{va'}$ → UNIT LOAD @ a'



$$\bar{m}_{va'} = 2.5 - x_a' \quad 0 \leq x_a' \leq 2.5$$

$$\begin{aligned} \delta_{va'} &= \frac{1}{EI} \int_0^{2.5} (62.5 - 20x)(2.5 - x) dx = \frac{1}{EI} \int_0^{2.5} (156.25 - 62.5x - 50x + 20x^2) dx \\ &= \frac{1}{EI} \int_0^{2.5} (156.25 - 112.5x + 20x^2) dx = \frac{1}{EI} \left[156.25x - 112.5 \frac{x^2}{2} + \frac{20x^3}{3} \right] \Big|_0^{2.5} \\ &= \frac{1}{EI} [143.2292] \end{aligned}$$

δ_{vb} ⇒ UNIT LOAD @ b



$$\bar{m}_{vb} = 5 - x_b \quad 0 \leq x_b \leq 5$$

$$\bar{m}_{vb'} = 2.5 - x_b' \quad 0 \leq x_b' \leq 2.5$$

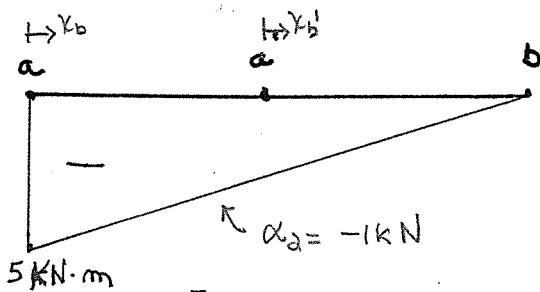
$$\delta_b = \frac{1}{EI} \int_0^5 M \bar{m}_b dx = \frac{1}{EI} \int_0^{2.5} (62.5 - 20x)(5 - x) dx + \frac{1}{EI} \int_0^{2.5} (12.5 - 5x)(2.5 - x) dx$$

AMPAD

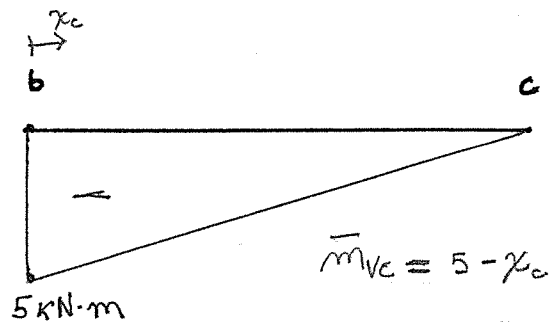
$$\begin{aligned} \bar{u}_b &= \frac{1}{EI} \int_0^{2.5} (312.5 - 162.5x + 20x^2) dx + \frac{1}{EI} \int_0^{2.5} (31.25 - 25x + 5x^2) dx \\ &= \frac{1}{EI} \left[312.5x - 162.5 \frac{x^2}{2} + 20 \frac{x^3}{3} \right] \Big|_0^{2.5} + \frac{1}{EI} \left[31.25x - 25 \frac{x^2}{2} + 5 \frac{x^3}{3} \right] \Big|_0^{2.5} \\ &= \frac{1}{EI} [377.6042] + \frac{1}{EI} [26.0417] \end{aligned}$$

$$\bar{u}_b = \frac{1}{EI} [403.6459]$$

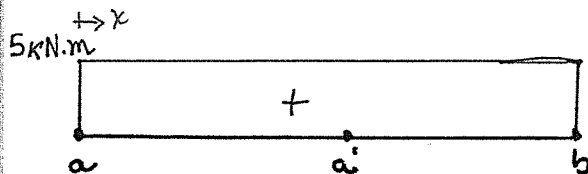
$\bar{v}_c \rightarrow$ UNIT LOAD @ C



$$\begin{aligned} \bar{m}_{vb} &= 5 - x_b & 0 \leq x_b \leq 2.5 \\ \bar{m}'_{vb} &= 2.5 - x'_b & 0 \leq x'_b \leq 2.5 \end{aligned}$$



$$0 \leq x_c \leq 5$$



$$\bar{T}_{ab} = 5 \quad V \cdot x [a, b]$$

$$\begin{aligned} \bar{v}_c &= \frac{1}{EI} \int_a^b M \bar{m} dx + \frac{1}{EI} \int_b^c M \bar{m} dx + \frac{1}{GJ} \int_a^b T \bar{T} dx \\ &= \bar{v}_b + \frac{1}{EI} \int_0^5 (12.5 - \frac{x^2}{2})(5-x) dx + \frac{1}{GJ} \int_0^5 (12.5)(5) dx \\ &= \bar{v}_b + \frac{1}{EI} \int_0^5 (62.5x - 12.5x - 5 \frac{x^2}{2} + \frac{x^3}{2}) dx + \frac{1}{GJ} \int_0^5 62.5 dx \end{aligned}$$

$$\begin{aligned} \delta_{vc} &= \delta_{vb} + \frac{1}{EI} \left(62.5x - 12.5 \frac{x^2}{2} - 5 \frac{x^3}{6} + \frac{x^4}{8} \right) \Big|_0^5 + \frac{1}{GJ} (62.5x) \Big|_0^5 \\ &= \frac{1}{EI} [403.6459] + \frac{1}{EI} [206.6458] + \frac{1}{GJ} [312.5] \end{aligned}$$

$$\delta_{vc} = \frac{1}{EI} [610.2917] + \frac{1}{GJ} [312.5]$$

SUMMARY

$$\delta_{va'} = \frac{1}{EI} (143.2292)$$

$$\delta_{vb} = \frac{1}{EI} (403.6459)$$

$$\delta_{vc} = \frac{1}{EI} (610.2917) + \frac{1}{GJ} [312.5]$$

EVALUATION OF E, I, G, J

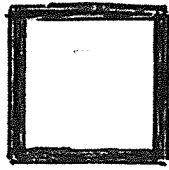
$$E = 200 \text{ GPa} = 2 \times 10^{11} \text{ Pa} = 2 \times 10^8 \text{ kN/m}^2$$

$$G = \frac{E}{2(1+\nu)} = \frac{2 \times 10^8}{2(1+0.3)} = 7.69230769 \times 10^7 \text{ kN/m}^2$$

I	W18x60	$I = 984 \text{ IN}^4 = 4.0957172 \times 10^{-4} \text{ m}^4$	$A = 17.6 \text{ IN}^2$
		$J = 2.17 \text{ IN}^4 = 9.0322219 \times 10^{-7} \text{ m}^4$	

$$\delta_{va'} = 0.001749 \text{ m} = 0.1749 \text{ cm} \qquad \delta_{vc} = 4.5052 \text{ mm}$$

$$\delta_{vb} = 0.004927 \text{ m} = 0.4927 \text{ cm}$$



HSS 10x10x1/2

$$I = 256 \text{ IN}^4 = 1.065555 \times 10^{-4} \text{ m}^4$$

$$J = 412 \text{ IN}^4 = 1.714873 \times 10^{-4} \text{ m}^4$$

$$A = 17.2 \text{ IN}^2$$

$$W = 62.46 \text{ lb}$$

$$\sqrt{v_a} = 0.006721 \text{ m} = 0.6721 \text{ cm}$$

$$\sqrt{v_b} = 0.018941 \text{ m} = 1.8941 \text{ cm}$$

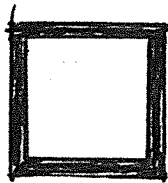
$$\sqrt{v_c} = 0.052327 \text{ m} = 5.2327 \text{ cm}$$

HYBRID



W18x60

b - b



HSS 10x10x1/2

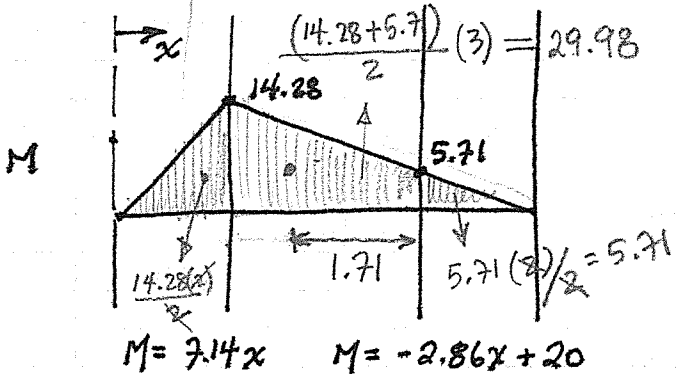
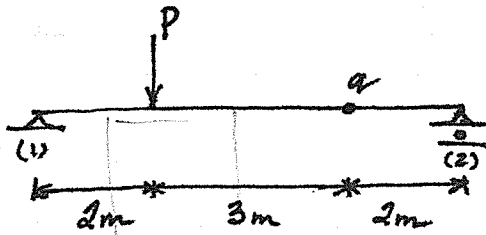
a - b

$$\sqrt{v_a} = 0.006721 \text{ m} = 0.6721 \text{ cm}$$

$$\sqrt{v_b} = 0.01841 \text{ m} = 1.8941 \text{ cm}$$

$$\sqrt{v_c} = 0.045153 \text{ m} = 4.5153 \text{ cm}$$

VIRTUAL WORK - COMPUTE DEFORMATIONS.



$$\delta_a =$$

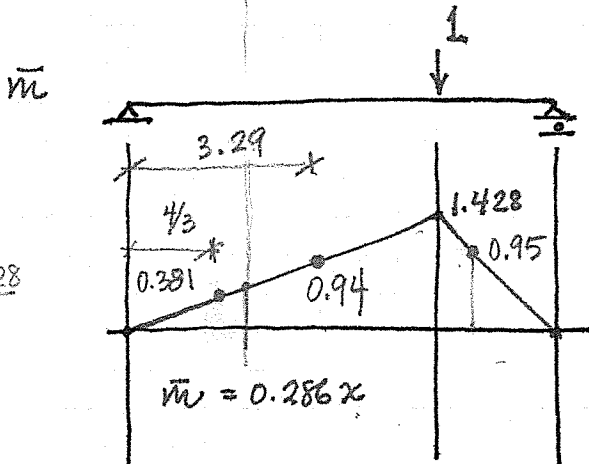
$$EI = \text{const.}$$

$$P = 10 \text{ kN.}$$

$$R_1 = \frac{5P}{7} = 0.71(10) = 7.14 \text{ kN}$$

$$R_2 = \frac{2P}{7} = 2.86 \text{ kN}$$

$$\int \frac{M \bar{m} dx}{EI} = A_M \cdot \bar{m}(\bar{x}_M)$$



$$R_1 = 0.286$$

$$R_2 = 0.714$$

$$\frac{1.428}{2} \cdot \left(\frac{4}{3}\right) =$$

$$EI \delta = \int_0^2 (7.14x)(0.286x) dx + \int_2^7 (-2.86x + 20)(0.286x) dx$$

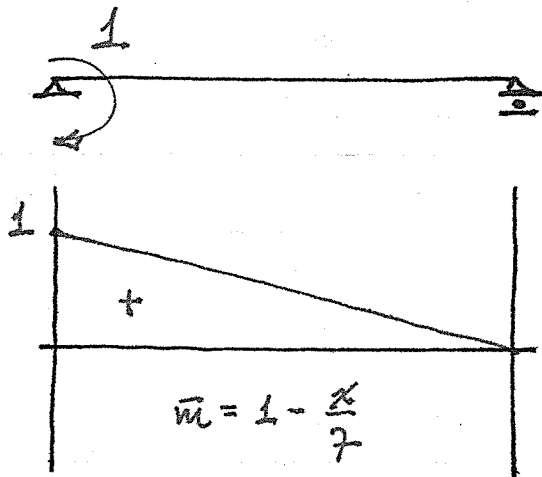
$(14.28)(0.381) \quad (29.98)(0.94)$
 $5.44 \quad 28.12$

$$+ \int_5^7 (-2.86x + 20)(-0.714x + 5) dx$$

$(5.71)(0.95) = 5.42$

$$5.445 + 28.16 + 5.419 = 39.02$$

Find rotation at support:



$$R_1 = -0.143$$

$$R_2 = 0.143$$

$$EI\theta = \int_0^2 (7.14x) \left(1 - \frac{x}{7}\right) dx + \int_2^7 (-2.86x + 20) \left(1 - \frac{x}{7}\right) dx$$

$$= A_m \cdot \bar{m}(\bar{x}_m) + A_m \bar{m}(\bar{x}_m)$$

$$= 14.28(0.81) + 35.7(0.476)$$

$$= 11.57 + 17.00$$

$$11.56 + 17.00$$